

# Mechanical Vibration

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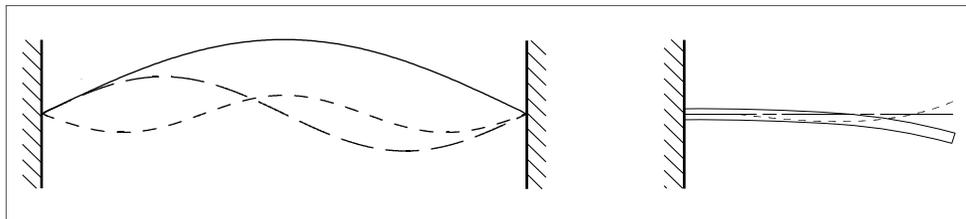
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## Introduction, basic definitions

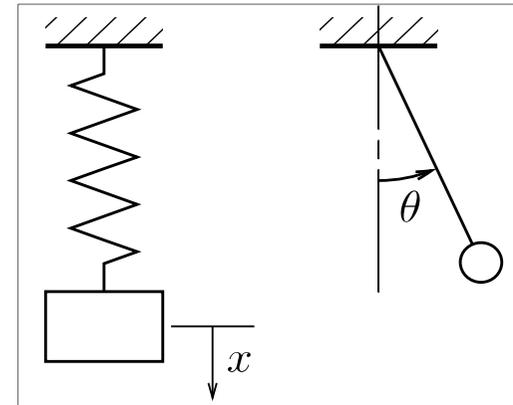
- Oscillatory process – alternate increases or decreases of physical quantities (displacements, velocities, accelerations)
- Oscillatory motion is periodic motion
- Vibration of turbine blades, vibration of machine tools, electrical oscillation, sound waves, vibration of engines, torsional vibration of shafts, vibration of automobiles etc.
- Mechanical vibration - mechanisms and machines, buildings, bridges, vehicles, aircrafts – cause mechanical failure
- Harmonic, periodic general motion

# Elastic elements

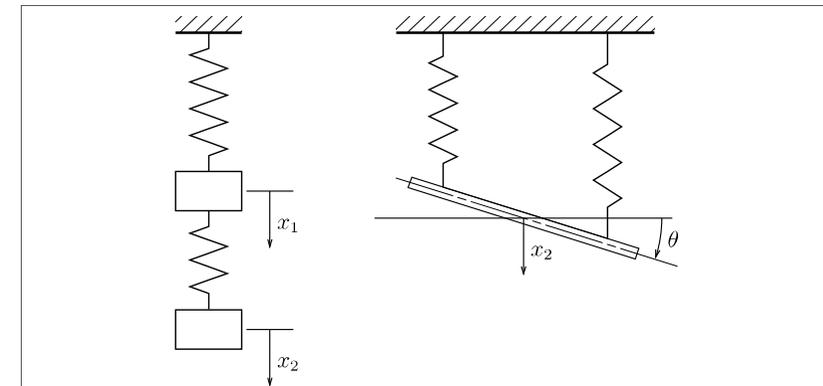
- Discrete elements (masses) + linear and torsional springs
- Continuous structural elements – beams and plates
- Number of degrees of freedom (DOF) – minimum number of coordinates



System with infinite number DOF



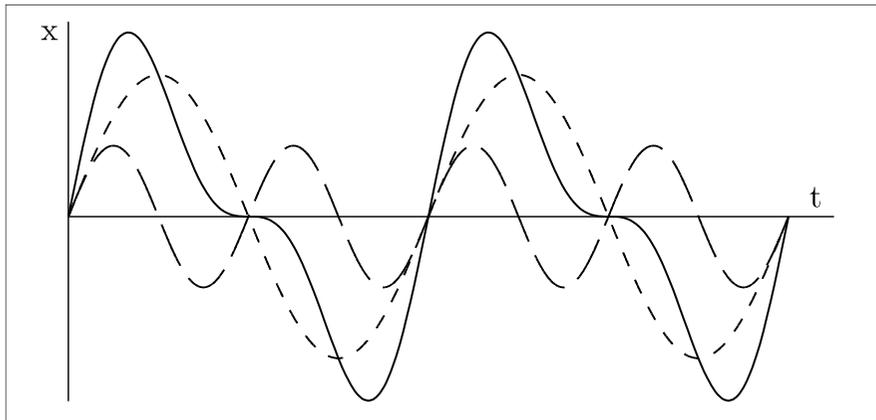
Single degree of freedom systems



Two degree of freedom systems

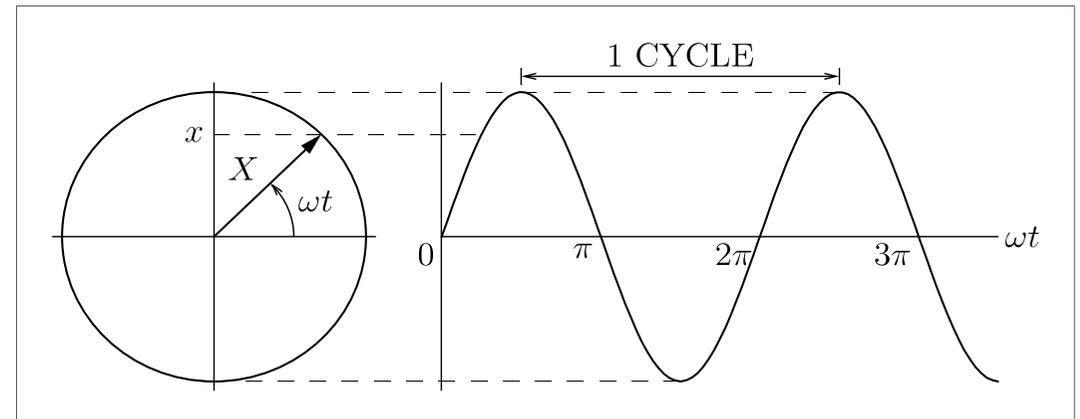
# Oscillatory motions

- Periodic motion with harmonic components



Periodic motion repeating itself after a certain time interval.

- Harmonic motion



The simplest form of periodic motion is harmonic motion – sin, cos

# Harmonic motion

- Displacement of harmonic motion is given:

$$x = X \sin(\omega t + \varphi)$$

- $x, x(t)$  ... displacement [m]
- $X$  ... amplitude of displacement [m]
- $(\omega t + \varphi)$  ... phase
- $\omega$  ... angular velocity [ $s^{-1}$ ]
- $T$  ... natural period of oscillation [s]
- $f$  ... frequency [ $s^{-1}$ ], [Hz] - Hertz
- $\varphi$  ... phase angle

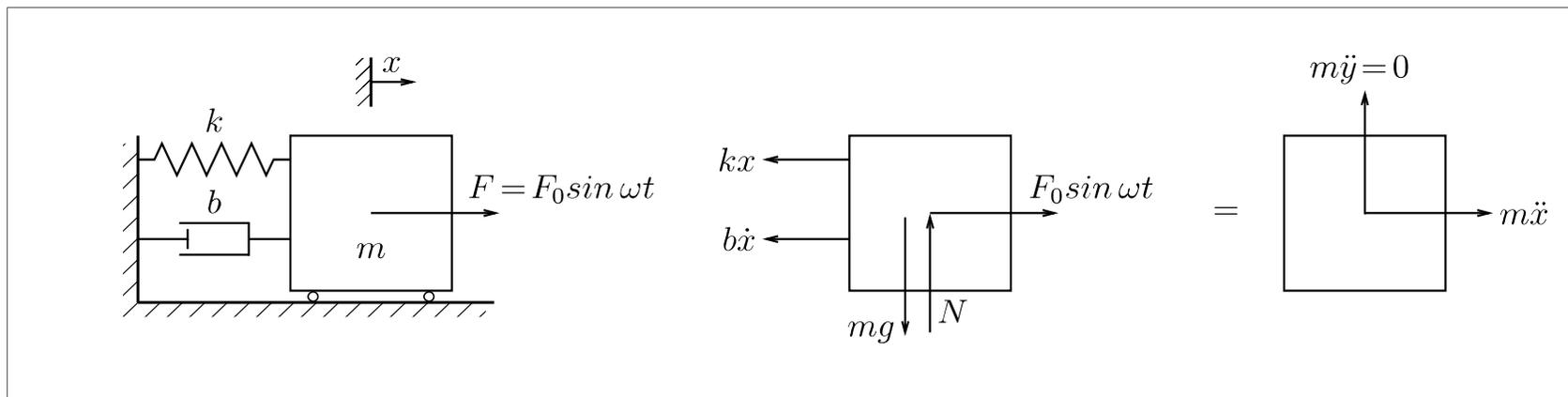
$$T = \frac{2\pi}{\omega}$$

$$\omega = 2\pi f$$

- velocity:  $\dot{x} = \frac{dx}{dt} = \omega X \cos(\omega t + \varphi)$
- $\omega X$  ... amplitude of velocity [ $\text{ms}^{-1}$ ]
- acceleration:  $\ddot{x} = \frac{d^2x}{dt^2} = -\omega^2 X \sin(\omega t + \varphi)$
- $-\omega^2 X$  ... amplitude of acceleration [ $\text{ms}^{-2}$ ]

# Simple degree of freedom systems

mass, spring, damper, harmonic excitation



Forcing function –  
harmonic excitation

$$m\ddot{x} + b\dot{x} + kx = F(t)$$

Damped free vibration

$$m\ddot{x} + b\dot{x} + kx = 0$$

Undamped free vibration

$$m\ddot{x} + kx = 0$$

# Simple degree of freedom systems

$$m\ddot{x} + b\dot{x} + kx = F(t)$$

$$F(t) = F_0 \sin \omega t$$

$m$  ... mass

$b$  ... (viscous) damping coefficient

$k$  ... stiffness coefficient

$F_0$  ... amplitudes of force

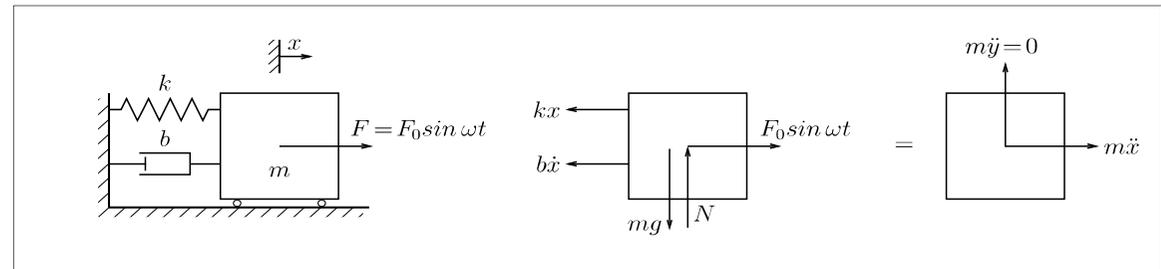
$\omega$  ... frequency of harmonic force

$b_{cr}$  ... critical damping coefficient

$\zeta$  ... damping factor

$$\Omega = \sqrt{\frac{k}{m}} \text{ ... natural (circular) frequency}$$

$$\eta = \frac{\omega}{\Omega} \text{ ... frequency ratio}$$



# Undamped free vibration

$$m\ddot{x} + kx = 0$$

Solution of the 2<sup>nd</sup> order differential equation

Assumed solution  $x(t) = Ae^{\lambda t}$

Characteristic equation:

$$m\lambda^2 + k = 0$$

$$\lambda^2 + \frac{k}{m} = 0$$

$$\lambda^2 = -\frac{k}{m}$$

$$\lambda_{1,2} = \pm i\sqrt{\frac{k}{m}}$$

$$\lambda_{1,2} = \pm i\Omega$$

$$\Omega = \sqrt{\frac{k}{m}}$$

Natural frequency  
of the single DOF systems

$$\ddot{x} + \frac{k}{m}x = 0 \quad \ddot{x} + \Omega^2 x = 0$$

$$x(t) = Ae^{i\Omega t} + Be^{-i\Omega t} \quad \text{free vibration}$$

Two arbitrary constants A and B, determined from initial conditions:

$$x(0) = x_0, \dot{x}(0) = v_0$$

$$\dot{x}(t) = i\Omega (Ae^{i\Omega t} - Be^{-i\Omega t})$$

$$x_0 = A + B$$

$$v_0 = i\Omega (A - B)$$

$$A = \frac{1}{2} \left( \frac{ix_0\Omega + v_0}{i\Omega} \right)$$

$$B = \frac{1}{2} \left( \frac{ix_0\Omega - v_0}{i\Omega} \right)$$

# Damped free vibration

$$m\ddot{x} + b\dot{x} + kx = 0$$

Solution of linear differential equation of 2nd order:

$$m\lambda^2 + b\lambda + k = 0$$

$$\lambda_{1,2} = \frac{-b}{2m} \pm \sqrt{b^2 - 4mk} \frac{1}{2m}$$

$$\lambda_{1,2} = \frac{-b}{2m} \pm i\sqrt{\frac{4mk}{4m^2} - \left(\frac{b}{2m}\right)^2} = \frac{-b}{2m} \pm i\sqrt{\frac{k}{m}} \sqrt{1 - \left(\frac{b}{2m\sqrt{\frac{k}{m}}}\right)^2}$$

$$\lambda_{1,2} = \frac{-b}{2m} \pm i\Omega\sqrt{1 - \left(\frac{b}{2\sqrt{km}}\right)^2} = \frac{-b}{2m} \pm i\Omega\sqrt{1 - \zeta^2}$$

$$\frac{b}{2\sqrt{km}} = \frac{b}{b_{CR}} = \zeta \quad \text{damping factor}$$

$$b_{CR} = 2\sqrt{km} \quad \text{critical damping coefficient}$$

$$\Omega\sqrt{1 - \zeta^2} = \Omega_D \quad \text{damped natural frequency}$$

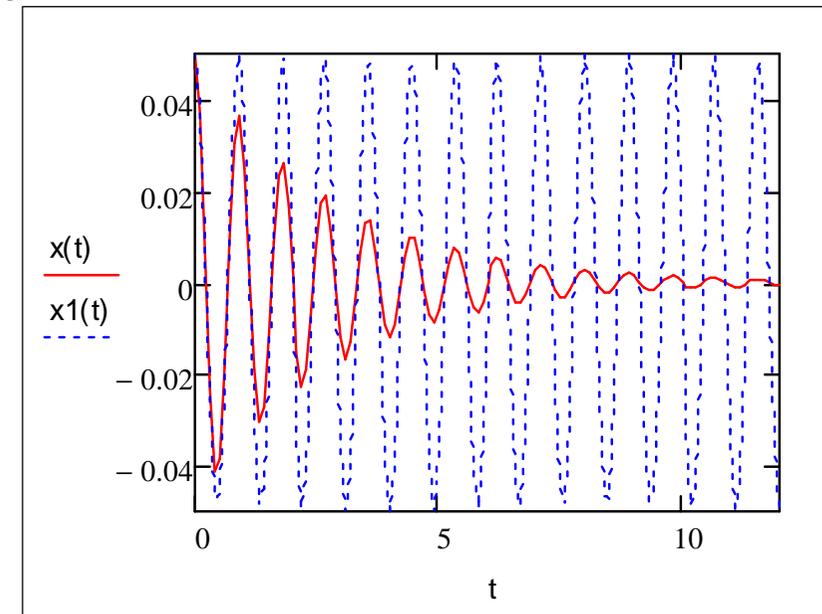
$$\lambda_{1,2} = \left(-\zeta \pm \sqrt{\zeta^2 - 1}\right)\Omega$$

$$x(t) = Ae^{(-\zeta + \sqrt{\zeta^2 - 1})\Omega t} + Be^{(-\zeta - \sqrt{\zeta^2 - 1})\Omega t}$$

$\zeta > 1$  overdamped system

$\zeta < 1$  underdamped system

$\zeta = 1$  critically damped system



Undamped --- and damped free vibration ---

# Damped free vibration

**Overdamped system:** displacement becomes the sum of two decaying exponentials with initial value of A+B, no vibration takes place, the body tends to creep back to the equilibrium position – APERIODIC MOTION (Fig.1)

$$\zeta > 1$$

$$x = Ae^{(-\zeta + \sqrt{\zeta^2 - 1})\Omega t} + Be^{(-\zeta - \sqrt{\zeta^2 - 1})\Omega t}$$

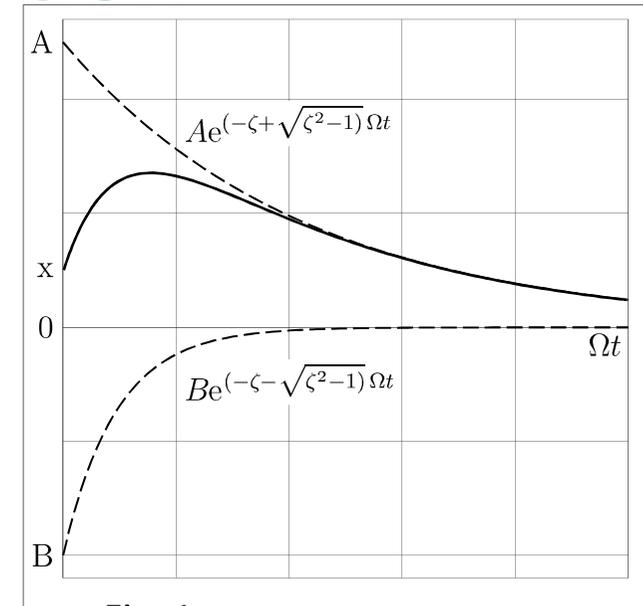


Fig. 1

**Underdamped system:** displacement is oscillatory with diminishing amplitude (Fig.2). Frequency of oscillation is less than that of the undamped case by the factor  $\sqrt{1-\zeta^2}$ .

$$\zeta < 1$$

$$\begin{aligned} x &= e^{-\zeta\Omega t} [C_1 e^{i\sqrt{1-\zeta^2}\Omega t} + C_2 e^{-i\sqrt{1-\zeta^2}\Omega t}] = \\ &= e^{-\zeta\Omega t} (A \cos \Omega_D t + B \sin \Omega_D t) = \\ &= Ce^{-\zeta\Omega t} \sin(\sqrt{1-\zeta^2}\Omega t + \gamma) \end{aligned}$$

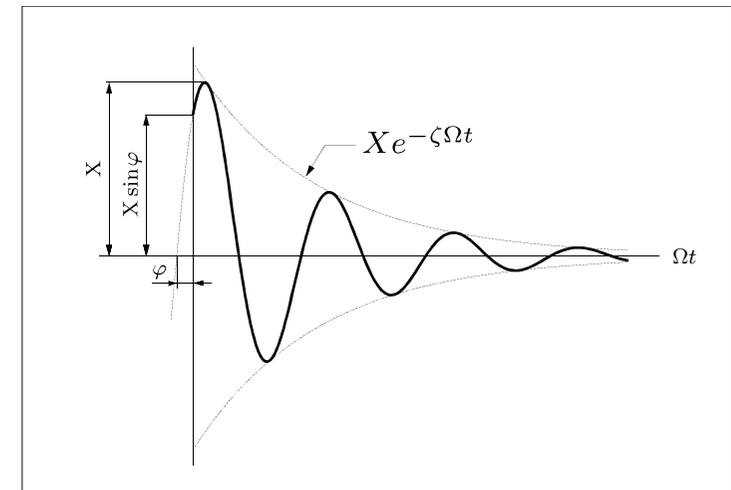


Fig. 2

**Critical damping:**

$$\zeta = 1$$

$$x = [A + Bt]e^{-\Omega t}$$

# Damped free vibration – Logarithmic decrement

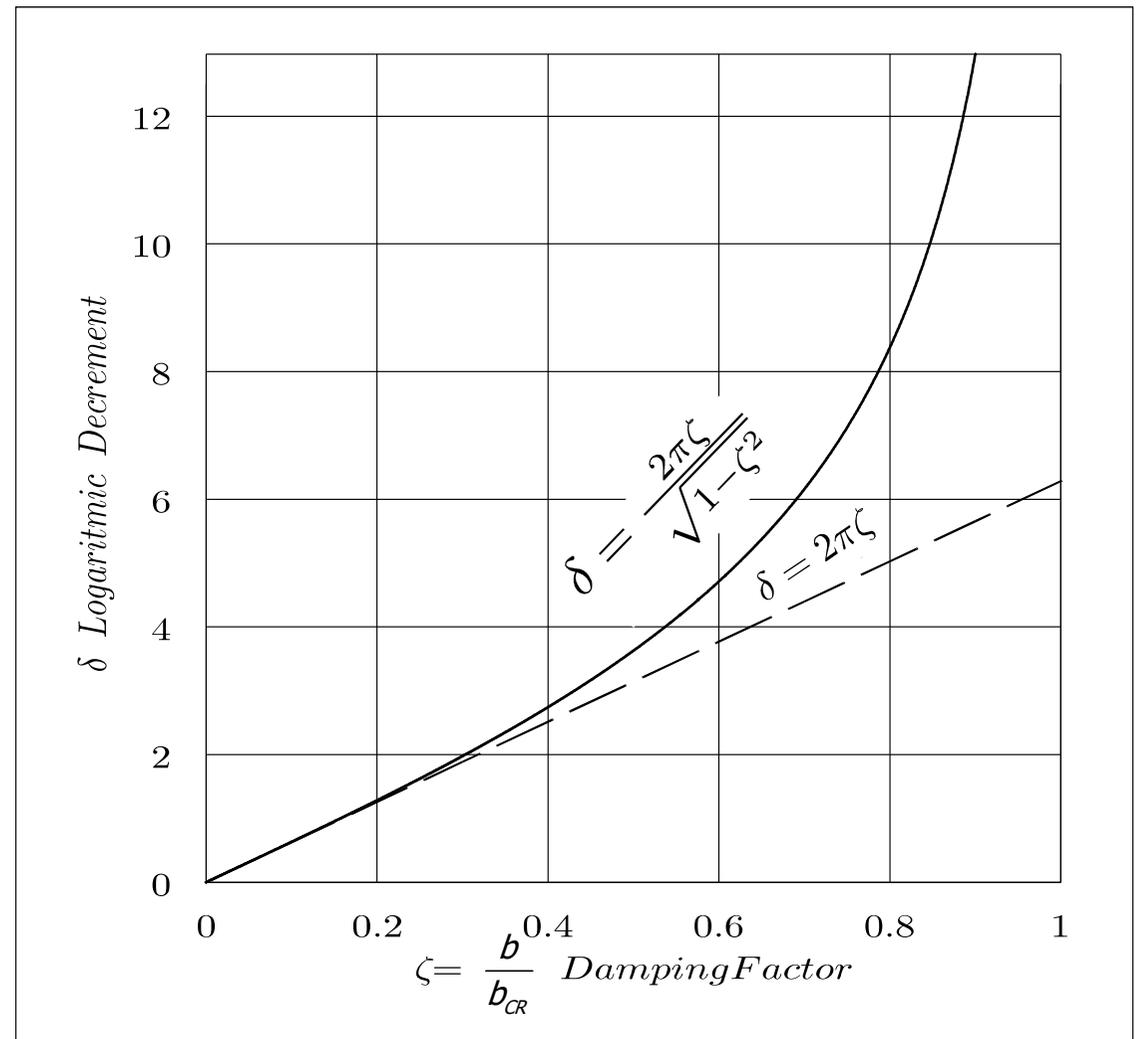
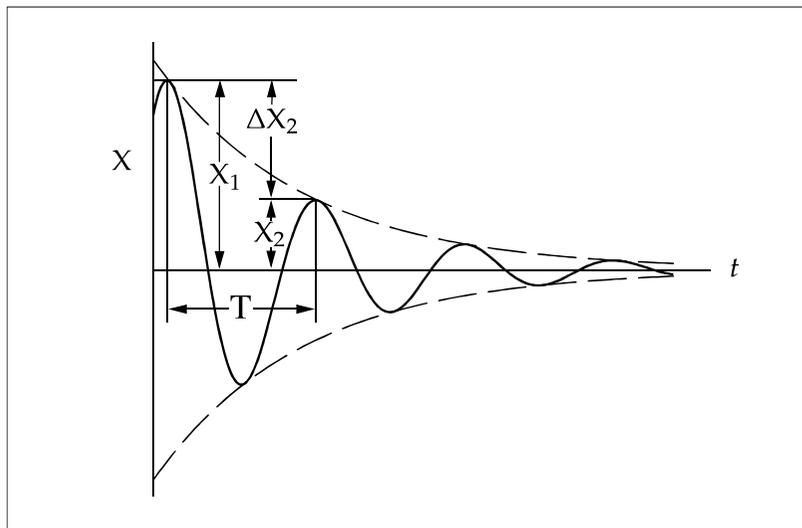
Natural logarithm of the ratio of any two amplitudes

$$\delta = \ln \frac{x_1}{x_2} = \ln \frac{x(t)}{x(t+T)} = \ln \frac{e^{-\zeta\Omega t} (A \cos \Omega_D t + B \sin \Omega_D t)}{e^{-\zeta\Omega(t+T)} (A \cos \Omega_D t + B \sin \Omega_D t)} =$$

$$= \ln \frac{1}{e^{-\zeta\Omega T}} = \zeta\Omega T = \zeta\Omega \frac{2\pi}{\Omega_T} = \frac{2\pi\zeta}{\sqrt{1-\zeta^2}}$$

$$\delta \doteq 2\pi\zeta \quad \text{for} \quad b \ll b_{CR}$$

$$\delta = \ln \frac{x(t)}{x(t+nT)} \doteq 2\pi n\zeta \Rightarrow \text{damping factor } \zeta = \frac{\delta}{2\pi n}$$



# Forced vibration, harmonic excitation

$$m\ddot{x} + b\dot{x} + kx = F(t)$$

$$m\ddot{x} + b\dot{x} + kx = F_0 \sin \omega t$$

## 1. Forced undamped vibration

- homogenous solution of equation

$$m\ddot{x} + kx = 0$$

- particular solution of equation

$$m\ddot{x} + kx = F_0 \sin \omega t$$

Solution of dif. equation with right side → steady state oscillation (response)

Assumed solution in the form of harmonic function:

$$x_p = a_1 \sin \omega t + a_2 \cos \omega t$$

$$\ddot{x}_p = -\omega^2 x_p = -\omega^2 (a_1 \sin \omega t + a_2 \cos \omega t)$$

$$\text{dif. equation: } (k - m\omega^2) a_1 \sin \omega t + (k - m\omega^2) a_2 \cos \omega t = F_0 \sin \omega t$$

Comparing of coefficients at function sin a cos on the both side of the equation → amplitude  $a_1$

$$a_1 = \frac{F_0}{k - m\omega^2}, a_2 = 0$$

# Forced vibration, harmonic excitation

Particular solution:  $x_p = \frac{F_0}{k - m\omega^2} \sin \omega t$

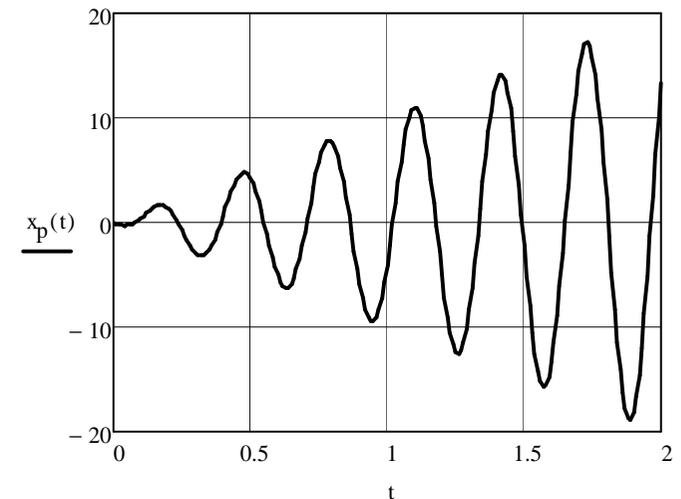
$$a = \frac{F_0}{k - m\omega^2} = \frac{F_0}{k} \frac{1}{1 - \frac{m}{k}\omega^2} = a_{ST} \frac{1}{1 - \eta^2}$$

$$a_{ST} = \frac{F_0}{k}$$

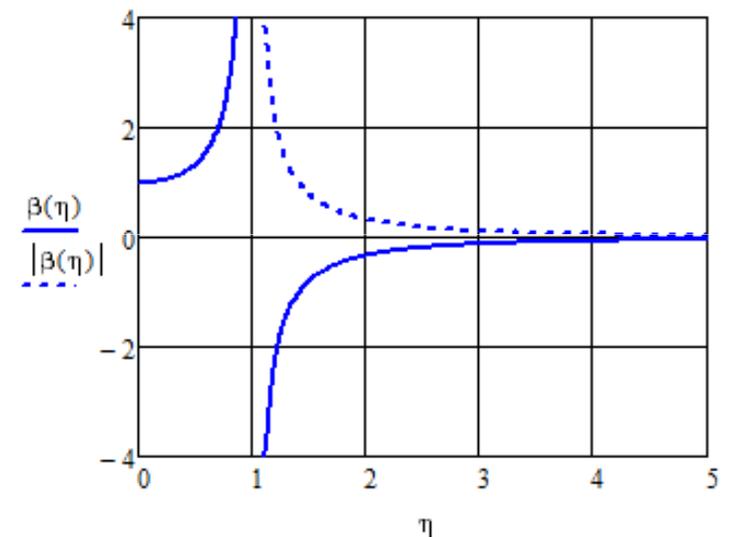
$$\frac{a}{a_{ST}} = \frac{1}{1 - \eta^2}$$

resonance – frequency of exciting force equals to natural frequency of the system

$$\eta = 1 \Rightarrow \omega = \Omega \Rightarrow \frac{a}{a_{ST}} \rightarrow \infty$$



Vibration of the single DOF system in resonance.



Resonance curve – undamped system

# Forced vibration

$$m\ddot{x} + b\dot{x} + kx = F(t)$$

$$m\ddot{x} + b\dot{x} + kx = F_0 \sin \omega t \quad (1)$$

harmonic force

1.  $m\ddot{x} + b\dot{x} + kx = 0$

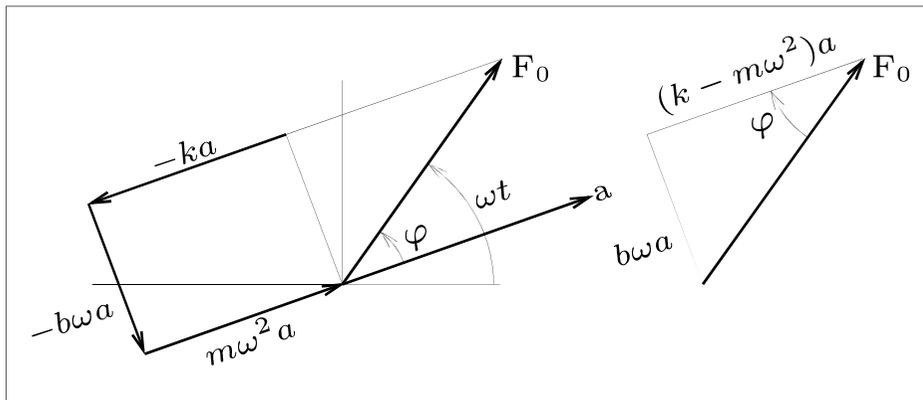
homogenous solution

2.  $m\ddot{x} + b\dot{x} + kx = F_0 \sin \omega t \rightarrow$  steady-state solution (response)

$$x = a \sin(\omega t - \varphi)$$

assumed solution of differential equation (1)

$$-ma\omega^2 \sin(\omega t - \varphi) + ba\omega \cos(\omega t - \varphi) + ka \sin(\omega t - \varphi) = F_0 \sin \omega t$$



Vectors' diagram

$$a = \frac{F_0}{\sqrt{(k - m\omega^2)^2 + (b\omega)^2}} = \frac{F_0}{k} \frac{1}{\sqrt{\left(1 - \frac{m}{k}\omega^2\right)^2 + \left(\frac{b}{k}\omega\right)^2}}$$

$$\tan \varphi = \frac{b\omega}{k - m\omega^2}$$

# Single degree of freedom system

Free damped vibration	Free undamped vibration	Forced damped vibration
$m\ddot{x} + b\dot{x} + kx = 0$	$m\ddot{x} + kx = 0$	$m\ddot{x} + b\dot{x} + kx = F(t)$
<p><b>Homogenous solution</b></p> $x_h = e^{-\zeta\Omega t} [Ae^{i\sqrt{1-\zeta^2}\Omega t} + Be^{-i\sqrt{1-\zeta^2}\Omega t}] =$ $= e^{-\zeta\Omega t} (A \cos \Omega_D t + B \sin \Omega_D t) =$ $= Ce^{-\zeta\Omega t} \sin(\sqrt{1-\zeta^2}\Omega t + \gamma)$	<p><b>Homogenous solution</b></p> $x_h = C_1 e^{i\Omega t} + C_2 e^{-i\Omega t}$ $x_h = A \cos \Omega t + B \sin \Omega t$ $x_h = C \sin(\Omega t + \gamma)$	$m\ddot{x} + b\dot{x} + kx = F_0 \sin \omega t$ $x = x_h + x_p$ $x_p = a \sin(\omega t - \varphi)$
<p><b>Initial conditions</b></p> $x(t_0) = x_0 \quad \dot{x}(t_0) = v_0$	<p><b>Initial conditions</b></p> $x(t_0) = x_0 \quad \dot{x}(t_0) = v_0$	<p><b>Amplitude of steady state oscillation, steady state response</b></p>
		$x(t) = e^{-\zeta\Omega t} [Ae^{i\sqrt{1-\zeta^2}\Omega t} + Be^{-i\sqrt{1-\zeta^2}\Omega t}]$ $+ \frac{F_0 \sin(\omega t - \varphi)}{\sqrt{(k - m\omega^2)^2 + (b\omega)^2}}$

# Forced vibration - magnification factor and phase angle

Solution of equation (1):  $x(t) = x_h + x_p$

$$x(t) = Ce^{-\zeta\Omega t} \sin(\sqrt{1-\zeta^2}\Omega t + \gamma) + \frac{F_0 \sin(\omega t - \varphi)}{\sqrt{(k - m\omega^2)^2 + (b\omega)^2}}$$

Values C a  $\gamma$  are derived from initial conditions.  
Amplitudes of steady-state oscillation:

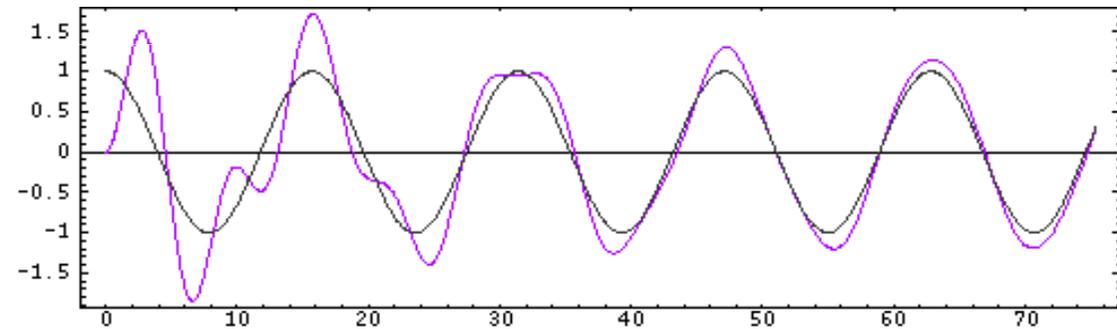
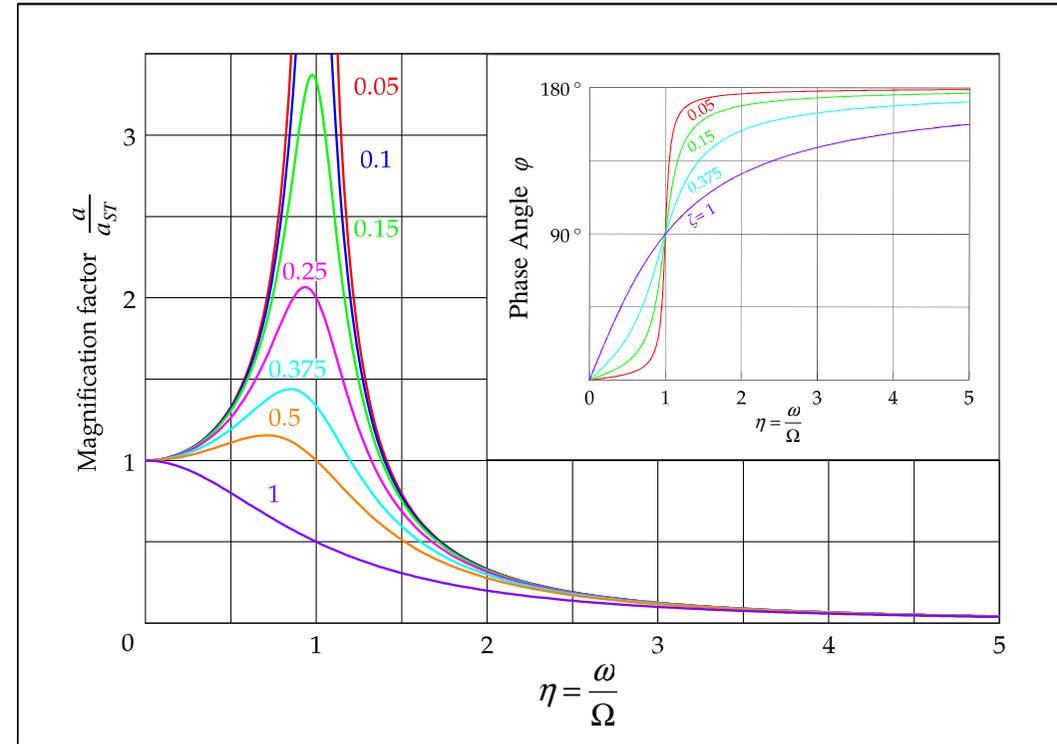
$$a = \frac{F_0}{\sqrt{(k - m\omega^2)^2 + (b\omega)^2}} = \frac{F_0}{k} \frac{1}{\sqrt{\left(1 - \frac{m\omega^2}{k}\right)^2 + \left(\frac{b}{k}\omega\right)^2}}$$

$\frac{F_0}{k} = a_{ST}$  statical deflection of the spring mass system under the action of steady force  $F_0$

$\eta = \frac{\omega}{\Omega}$  frequency ratio

$$\frac{a}{a_{ST}} = \frac{1}{\sqrt{(1 - \eta^2)^2 + (2\zeta\eta)^2}} \quad \text{magnification factor}$$

$$\varphi = \arctan \frac{2\zeta\eta}{1 - \eta^2} \quad \text{phase angle}$$

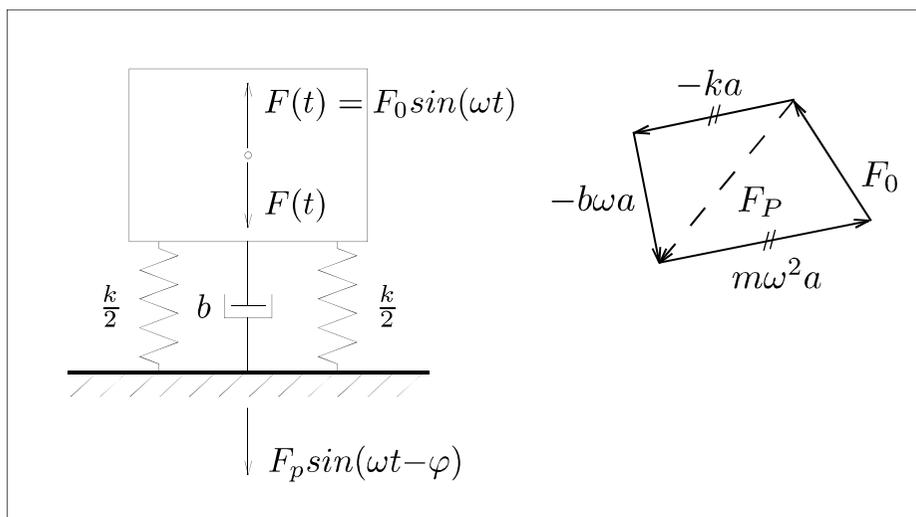


Transient motion, under resonance

# Vibration isolation and transmissibility

Machine or engines rigidly attached to a supporting structure, vibration is transmitted directly to the support (often undesirable vibration). Disturbing source must be isolated. Force is transmitted through springs and damper.

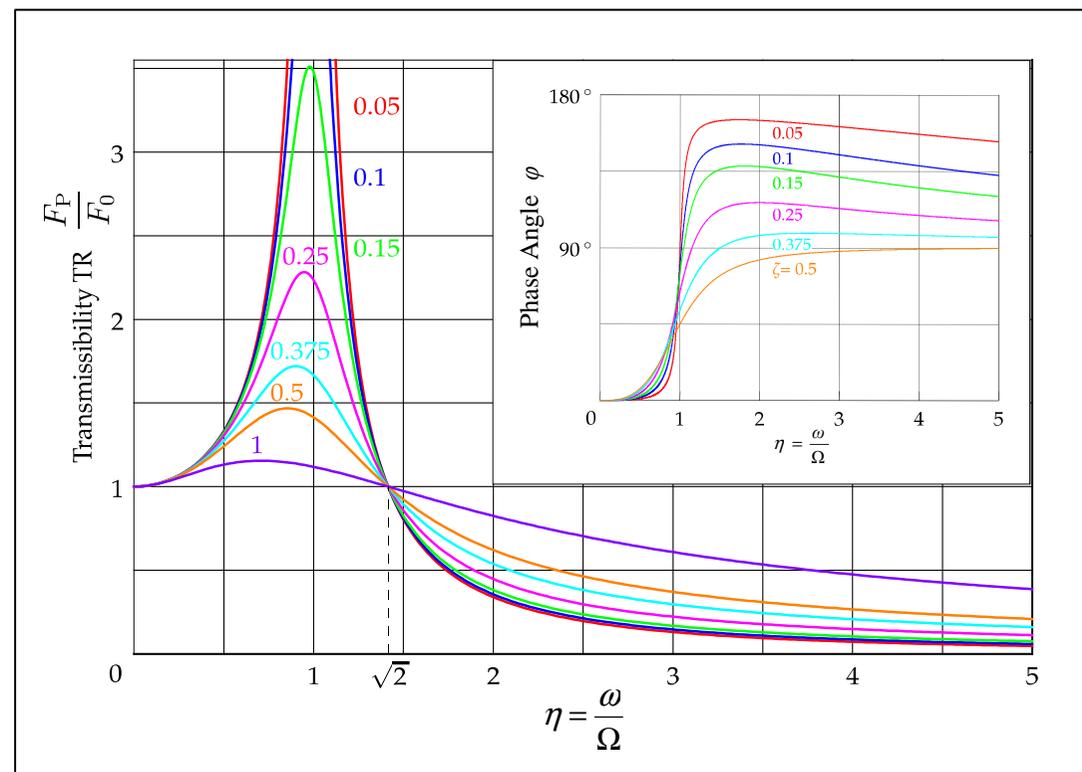
- a)  $\omega = \text{const.} \rightarrow \eta = \text{const.} \quad \omega \gg \Omega, \eta \gg 1$ , small damping factor  $\zeta$
- b)  $\omega \neq \text{const.} \rightarrow$  isolation, support with damping



$$F_P = \sqrt{(-ka)^2 + (b\omega a)^2} = a\sqrt{k^2 + (b\omega)^2}$$

$$= F_0 \sqrt{\frac{k^2 + (b\omega)^2}{(k - m\omega^2)^2 + (b\omega)^2}}$$

$$\frac{F_P}{F_0} = \sqrt{\frac{1 + (2\xi\eta)^2}{(1 - \eta^2)^2 + (2\xi\eta)^2}}$$

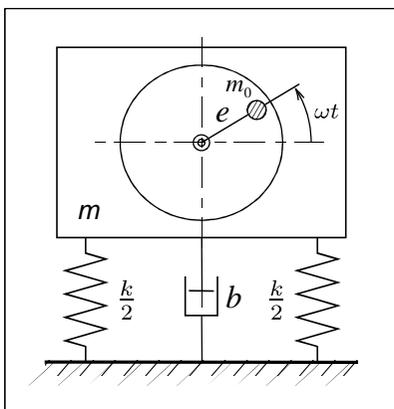


# Rotating unbalance

Rotating unbalance systems (gears, wheels, shafts disks which are not perfectly uniform, produce unbalance force which cause excessive vibrations.

$m_0$  ... unbalance mass

$e$  ... eccentricity

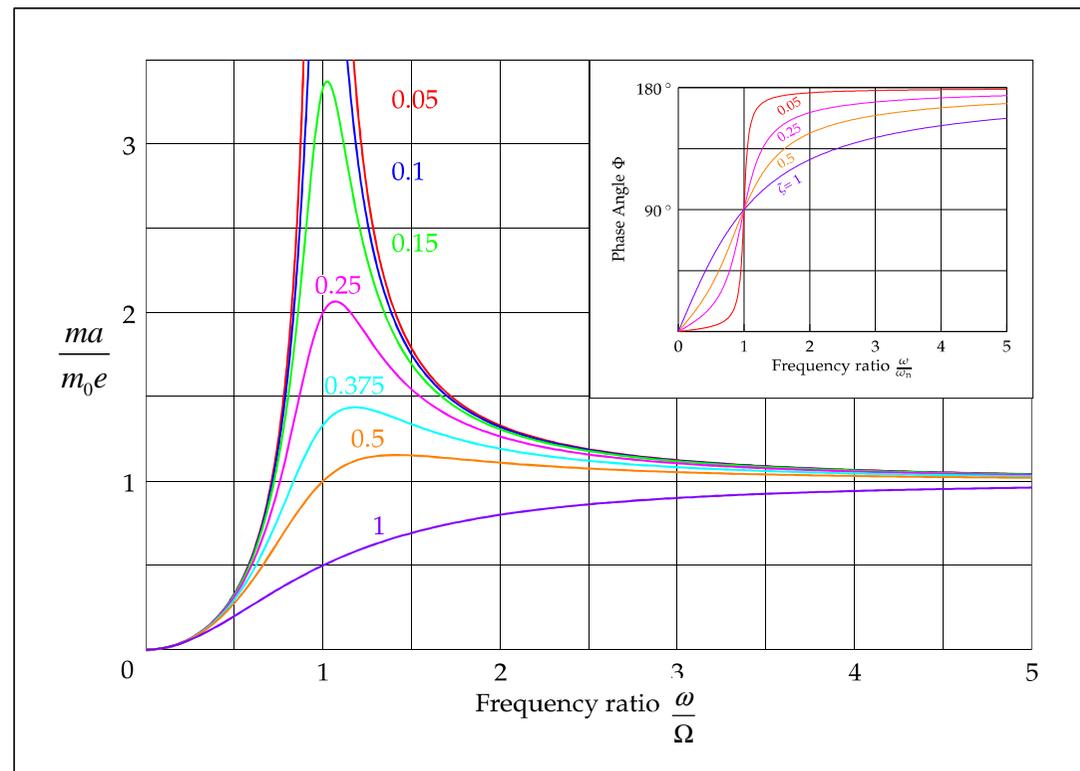


$$m\ddot{x} + b\dot{x} + kx = F(t)$$

$$m\ddot{x} + b\dot{x} + kx = m_0 e \omega^2 \sin \omega t$$

$$\frac{a}{m/m_0 e} = \frac{\eta^2}{\sqrt{(1-\eta^2)^2 + (2\zeta\eta)^2}}$$

$$\tan \varphi = \frac{b\omega}{\sqrt{k - m\omega^2}} = \frac{2\zeta\eta}{1-\eta^2}$$



# Base motion, relative motion

Forced vibration of mechanical systems can be caused by the support motion (vehicles, aircrafts and ships)

harmonic motion  $y = y_0 e^{i\omega t}$

$$m\ddot{x} + b(\dot{x} - \dot{y}) + k(x - y) = 0$$

$$m\ddot{x}_r + b\dot{x}_r + kx_r = -m\ddot{y}$$

$$m\ddot{x}_r + b\dot{x}_r + kx_r = m\omega^2 y_0 e^{i\omega t}$$

solution

$$x_p = a_r e^{i(\omega t - \varphi)}$$

$$a_r = \frac{m\omega^2 y_0}{\sqrt{(k - m\omega^2)^2 + (b\omega)^2}}$$

$$\frac{a_r}{y_0} = \frac{\eta^2}{\sqrt{(1 - \eta^2)^2 + (2\zeta\eta)^2}} \quad \tan \varphi = \frac{2\zeta\eta}{1 - \eta^2}$$

